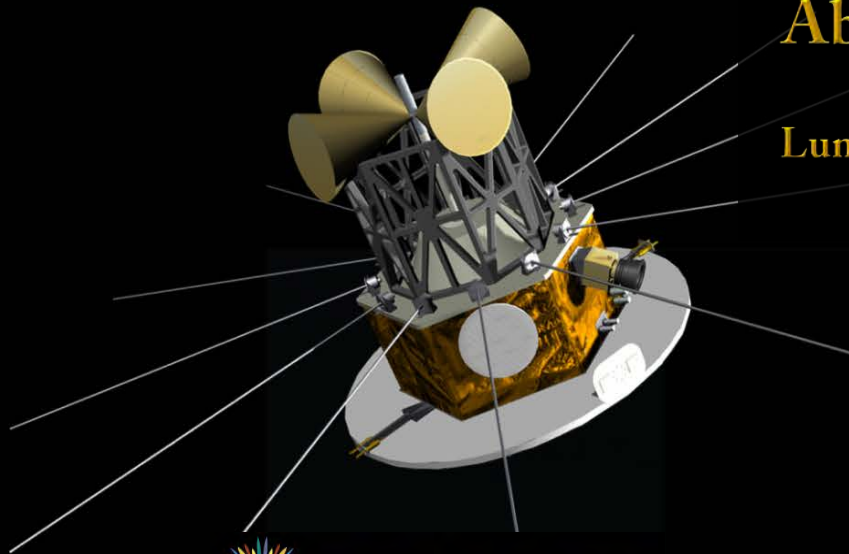


DARE

DARK AGES RADIO EXPLORER



Abhirup Datta for the DARE Team
University of Colorado Boulder,
Lunar University Network for Astrophysics Research,
& NASA Lunar Science Institute



DARE PROJECT TEAM

Principal Investigator:
Jack Burns, U. Colorado

Deputy Principal Investigator:
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Project Manager:
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Deputy Project Manager:
Jill Bauman, ARC

Spacecraft PM:
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Ball Aerospace

Instrument Manager:
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Collaborator: Michael Bicay, ARC

Science Co-Investigators

Stuart Bale, UC Berkeley

Judd Bowman, Arizona State Univ.

Richard Bradley, Natl. Radio Astronomy Obsv.

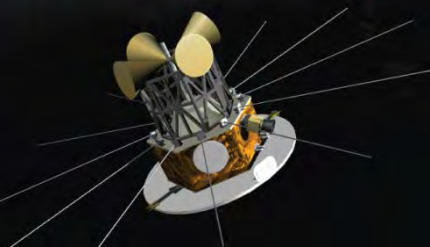
Christopher Carilli, Natl. Radio Astronomy Obsv.

Steven Furlanetto, UCLA

Geraint Harker & Abhi Datta, U. Colorado

Abraham Loeb, Harvard University

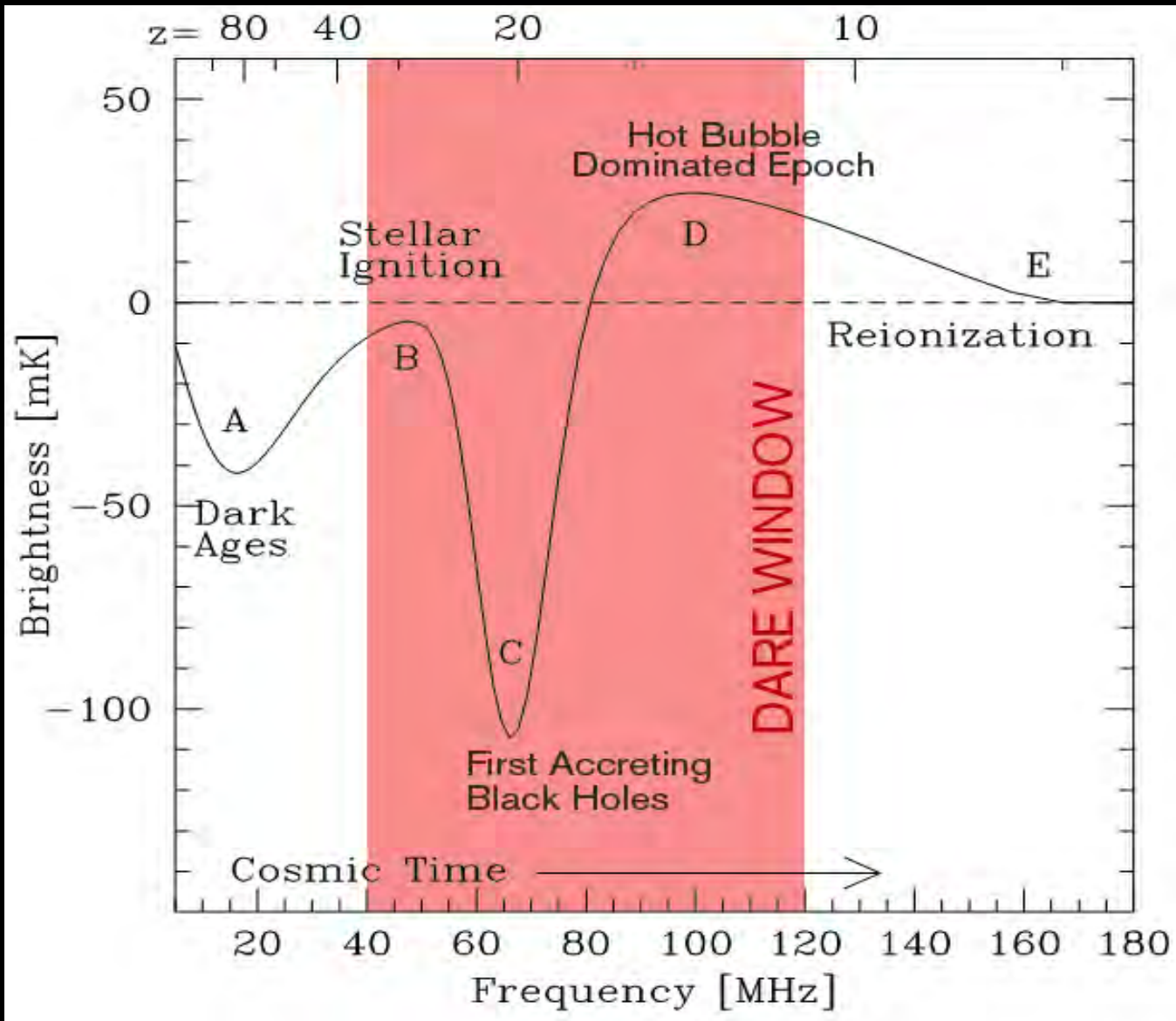
Jonathan Pritchard, Harvard-Smithsonian
Center for Astrophysics



PARTNERSHIPS

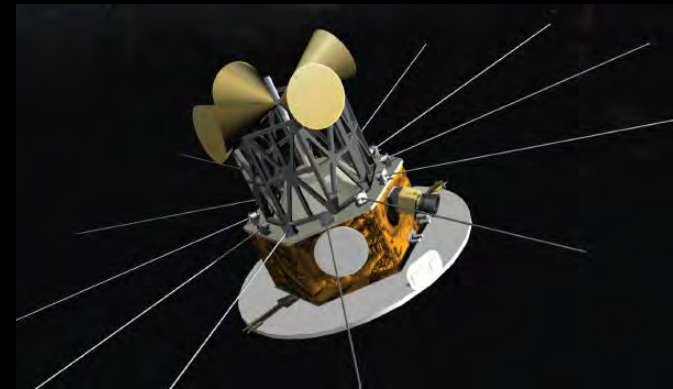
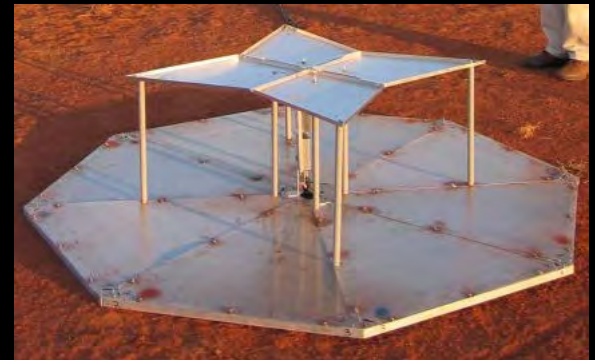


DARE will focus on determining or constraining *Turning Points B, C, D*



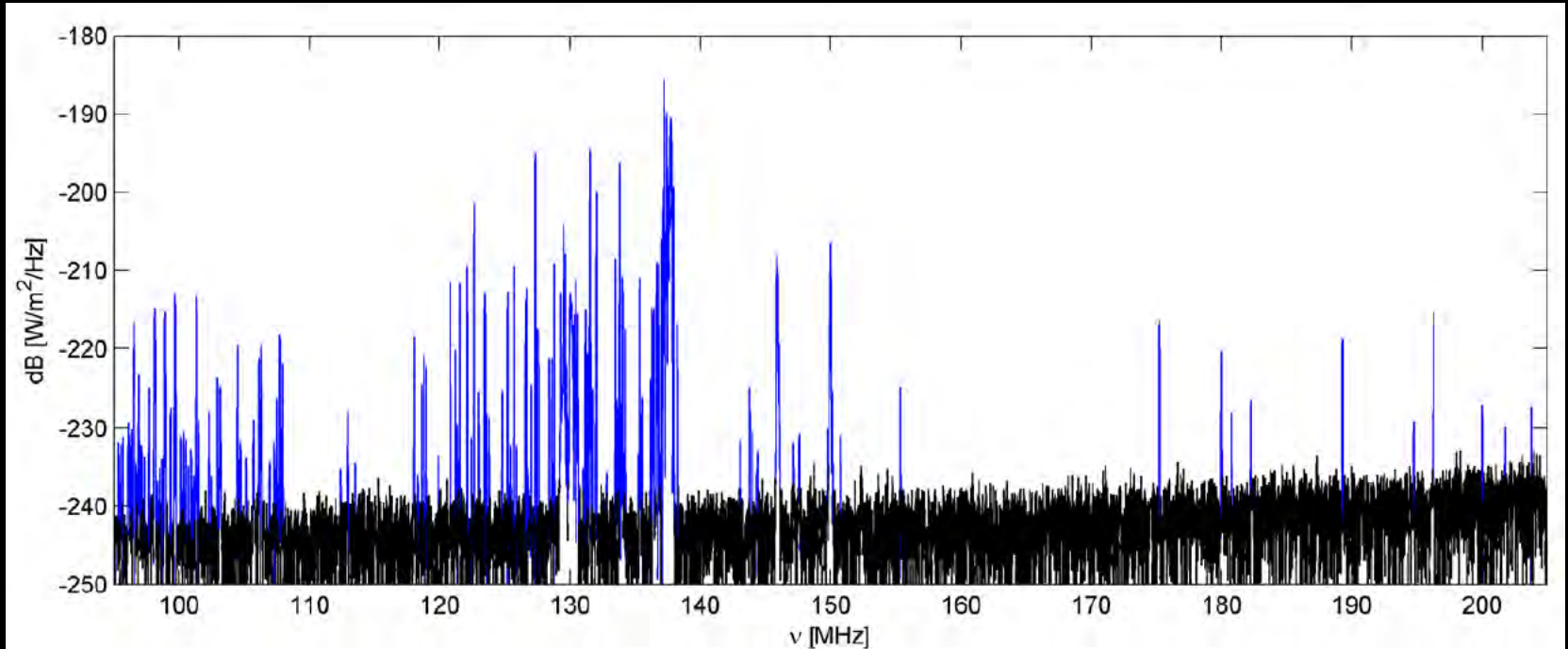
Lessons from EDGES

- 10^9 - 10^{10} dynamic range difficult because of RFI => A/D converters need high bit-depths & be highly linear. Susceptible to internal clock stability errors & digital noise.
- Multipath reflections=> complex spectral interference.
- Complex environment makes transferring instrument response function from lab impossible.
- Ionosphere adds significant noise at <80 MHz.

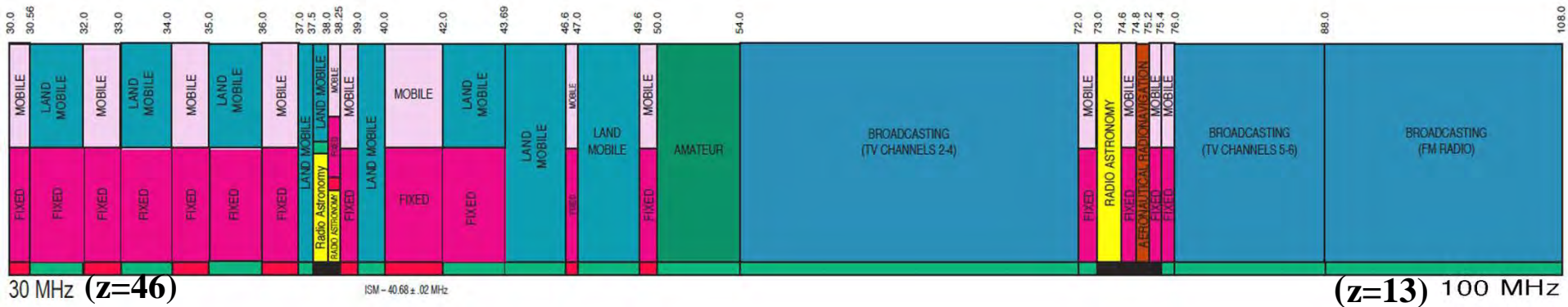


Analogous to why
COBE went
to space!

Radio Frequency Interference in Western Australia



Lunar Advantage: No Interference!



Destination: Moon

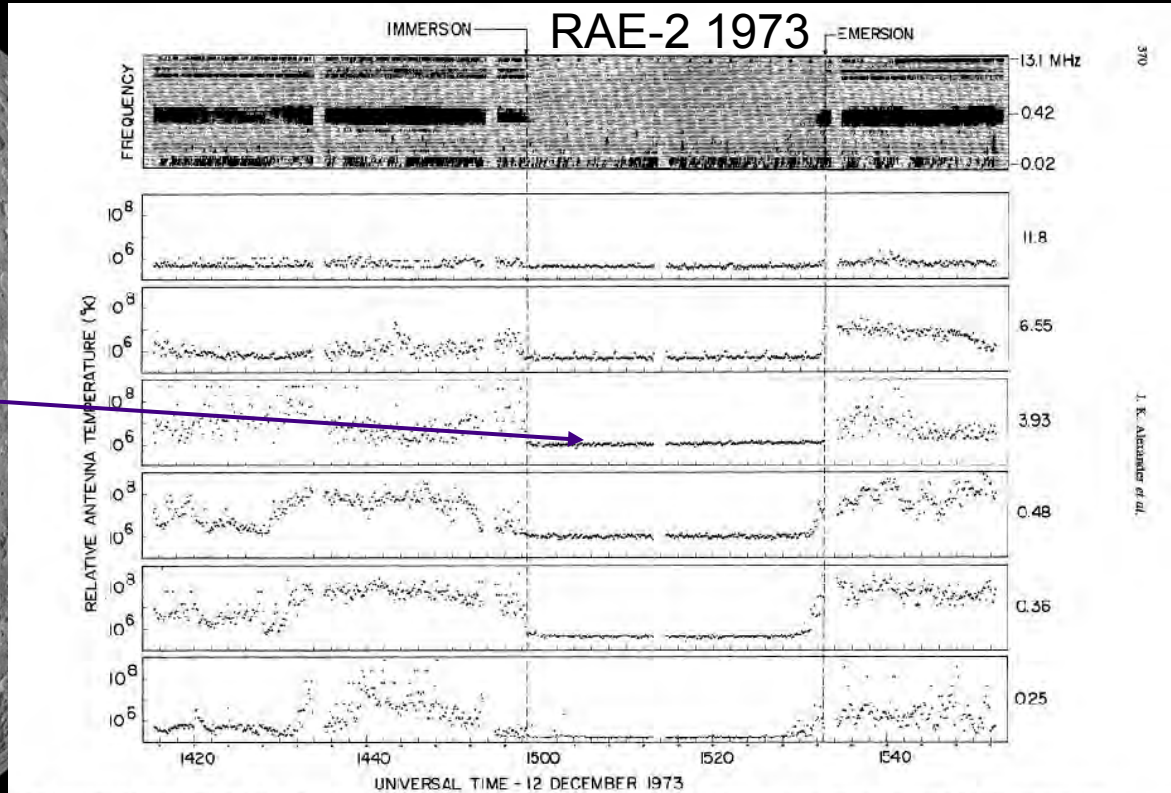
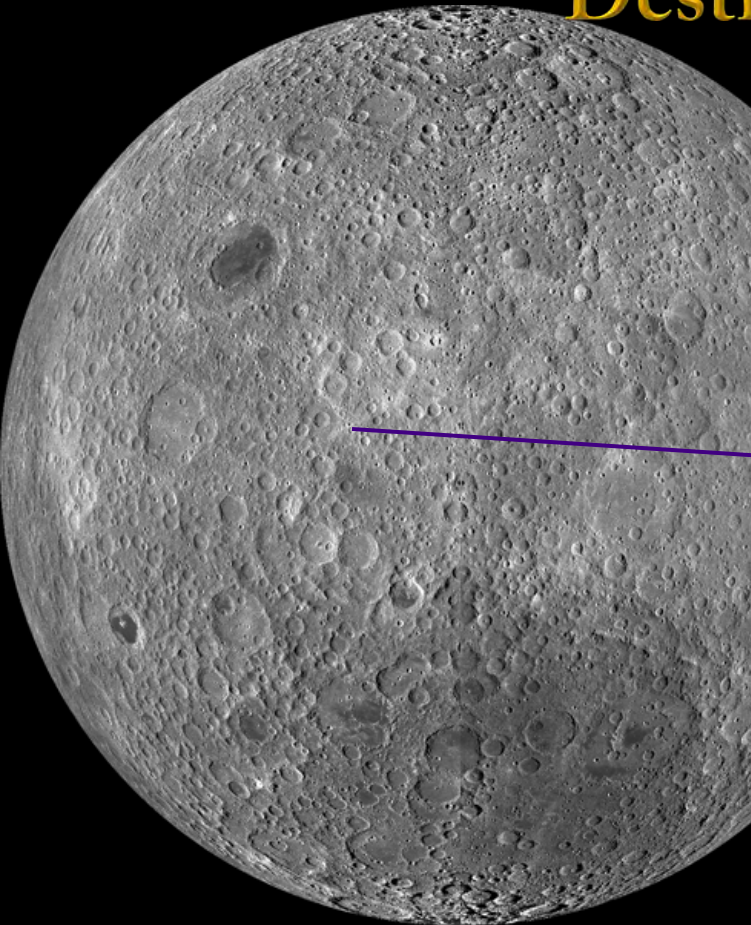


Fig. 5 Example of a lunar occultation of the Earth as observed with the upper-V burst receiver. The top frame is a computer-generated dynamic spectrum; the other plots display intensity vs. time variations at frequencies where terrestrial noise levels are often observed. The 80-s data gaps which occur every 20 m. are at times when in-flight calibrations occur. The short noise pulses observed every 144 s at the highest frequencies during the occultation period are due to weak interference from the Ryle-Vonberg receiver local oscillator on occasions when both the receiver and the burst receiver are tuned to the same frequency.

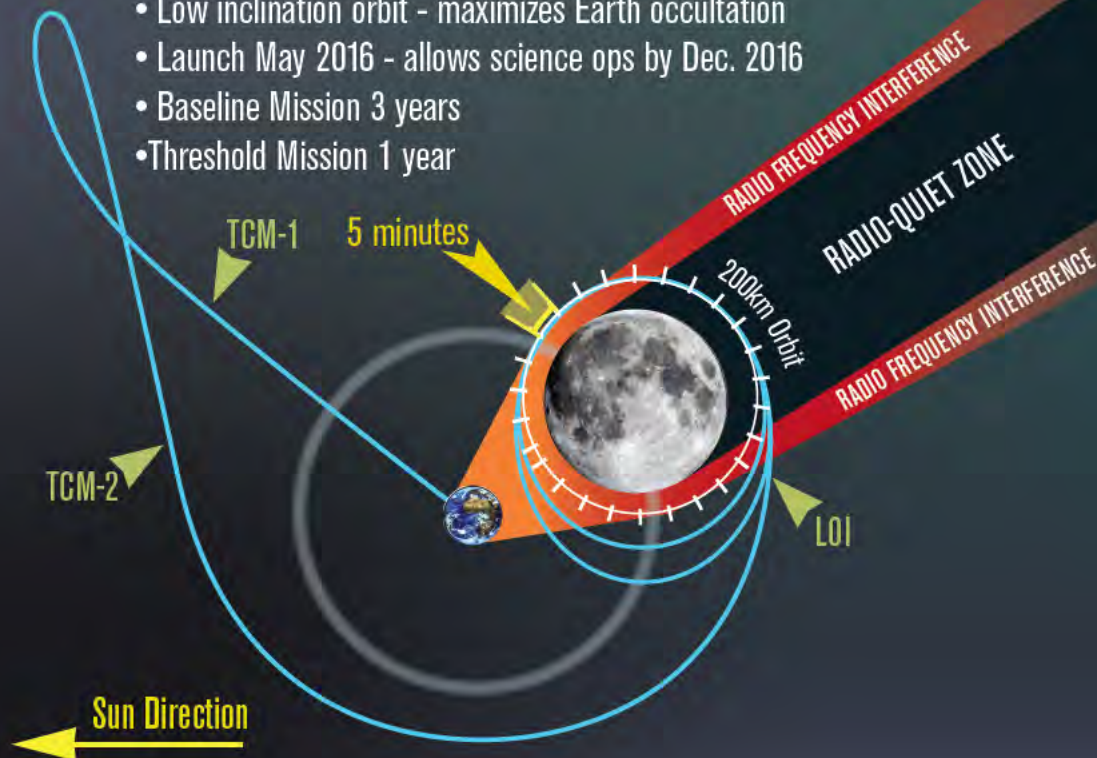
DARE's Biggest Challenge:

Foregrounds

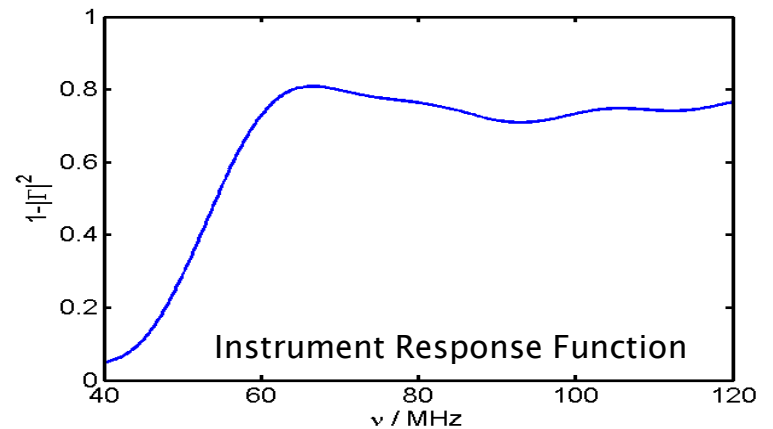
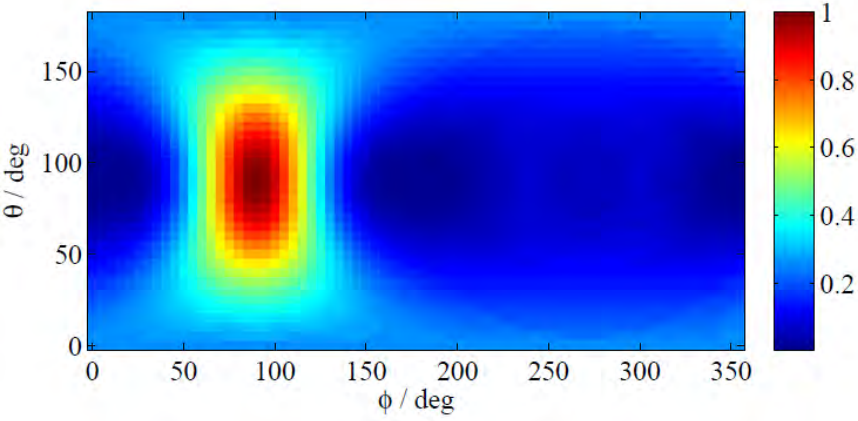
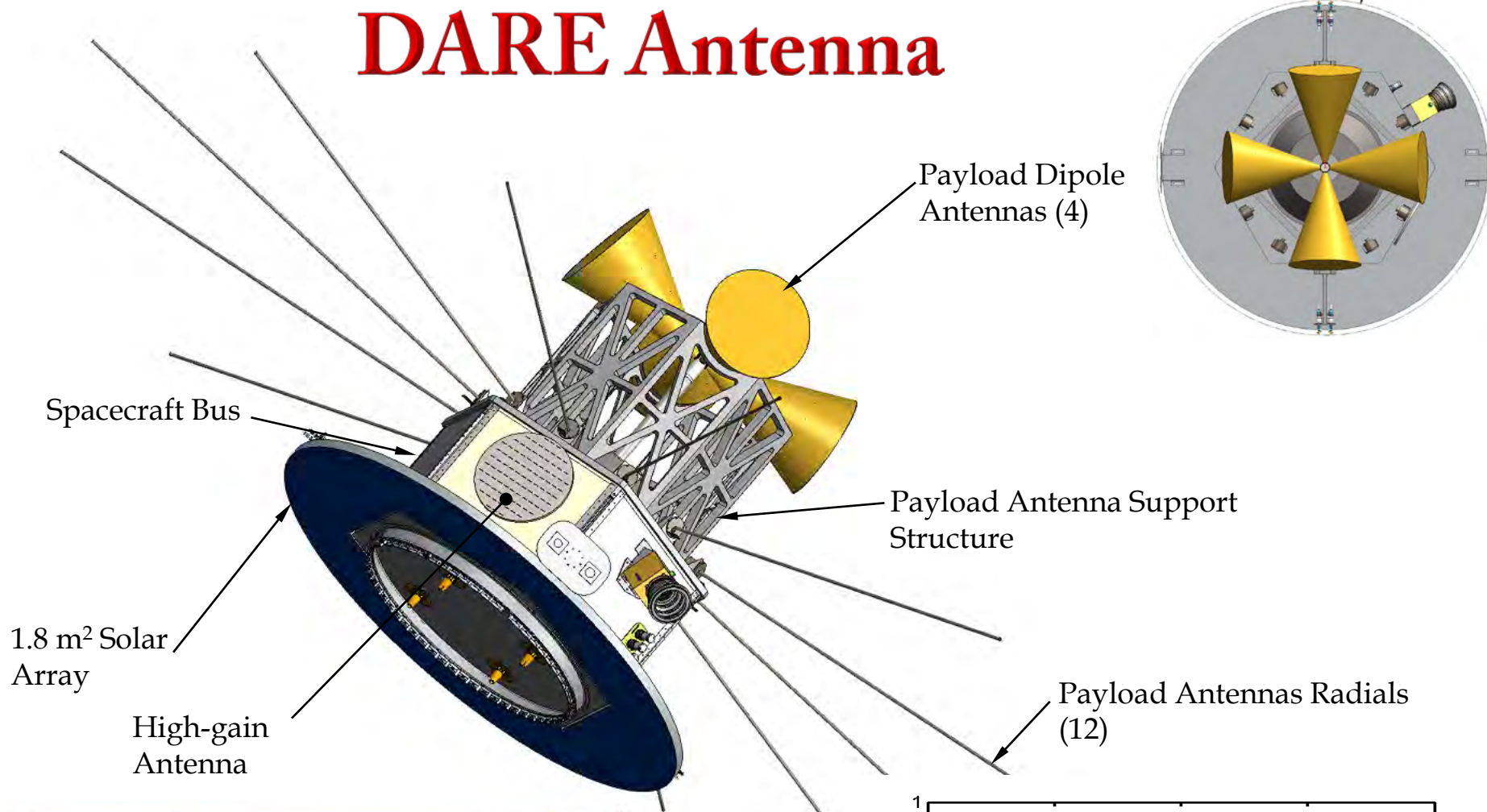
Highest foreground (RFI) eliminated by being above lunar farside!

DARE's Key Mission Design Features:

- Weak Stability Boundary (WSB) trajectory - requires less ΔV for LOI and allows a flexible launch date
- Equatorial, 200km mean orbit altitude - long-period stability
- Low inclination orbit - maximizes Earth occultation
- Launch May 2016 - allows science ops by Dec. 2016
- Baseline Mission 3 years
- Threshold Mission 1 year

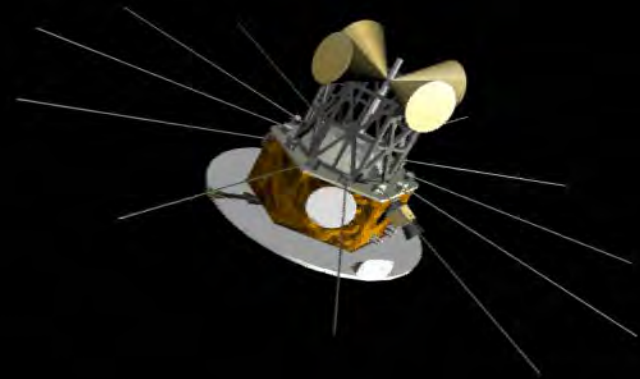


DARE Antenna



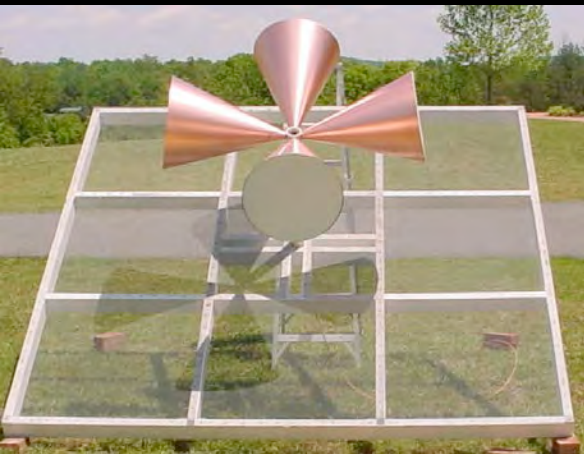
DARE Status & Timeline

- DARE was proposed as a Explorer Mission in February, 2011.
- Mission was not accepted for Phase A study— but very high referee ratings.
- DARE Engineering prototype has been developed (NRAO and JPL).
- Instrument Verification Program includes the initial field tests in Green Bank, WV (Feb-Mar, 2012) as well as the DARE-ground experiment in Western Australia (Mar, 2012 onwards).
- Results from these experiments will be critical in re-proposing DARE for a SMEX mission in late 2013.

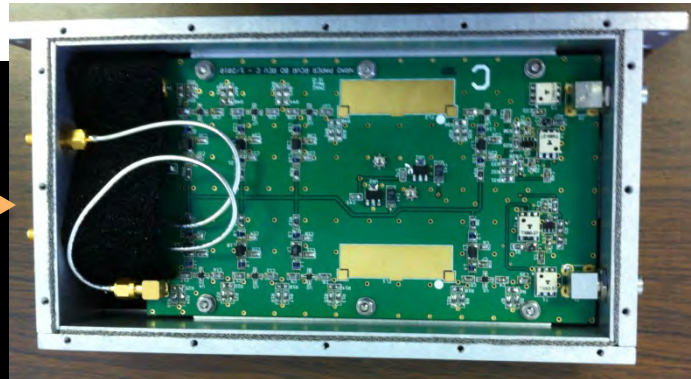
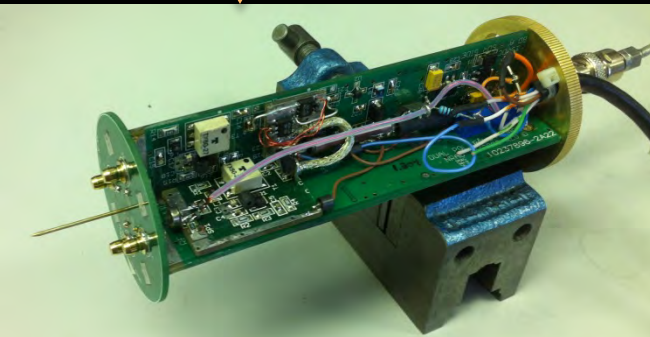
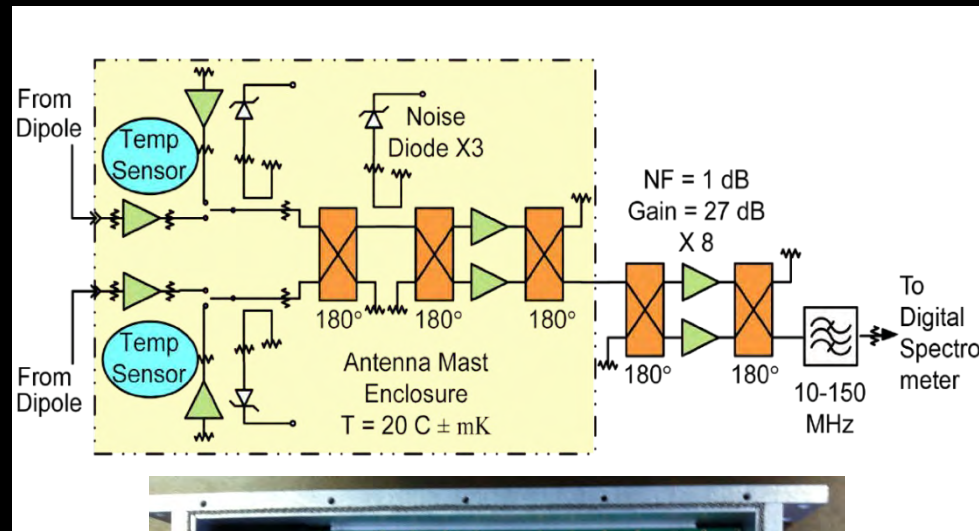


DARE Engineering Prototype: Components

- DARE will operate at low radio frequencies between 40-120 MHz
- Components of all three subsystems (antenna, receiver and spectrometer) are at TRL ≥ 6
- Instrument Verification Program underway to have the integrated instrument at TRL 6



Antenna + BALUN (NRAO)



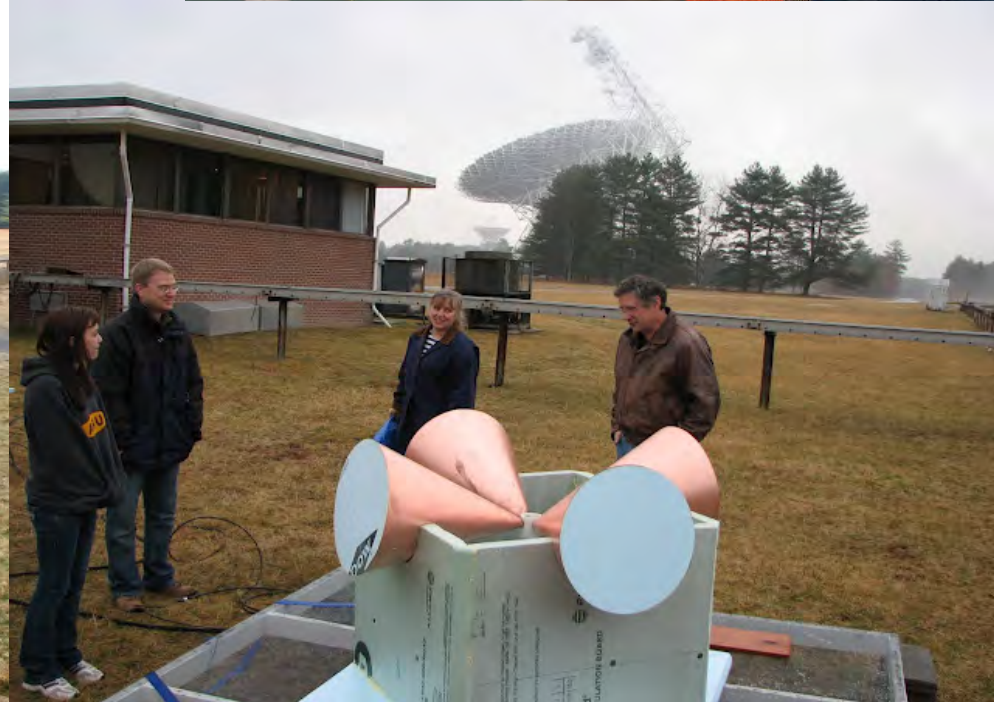
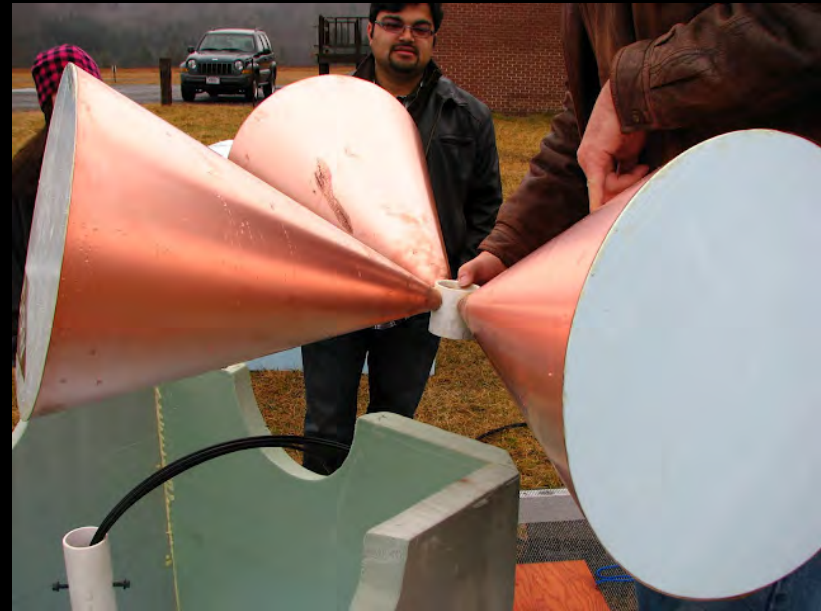
Front End Receiver (NRAO)



Digital Spectrometer (JPL)

Green Bank field tests

- DARE Engineering prototype was deployed at the NRAO site in Green Bank, WV.
- Recorded data for about 2 weeks.
- Initial field tests validated the performance of three stages of DARE instrument: antenna, front-end and digital spectrometer.



Initial Calibration

- $P_{OFF} = g(T_{Load} + T_{Rcvr})(1 + n_0)$
- $P_{ON} = g(T_{Ant} + T_{Rcvr})(1 + n_1)$

■ Equating these two we get :

$$T_{Ant} = \frac{P_{ON}}{P_{OFF}} (T_{Load} + T_{Rcvr}) - T_{Rcvr}$$

■ To the first order of approximations :

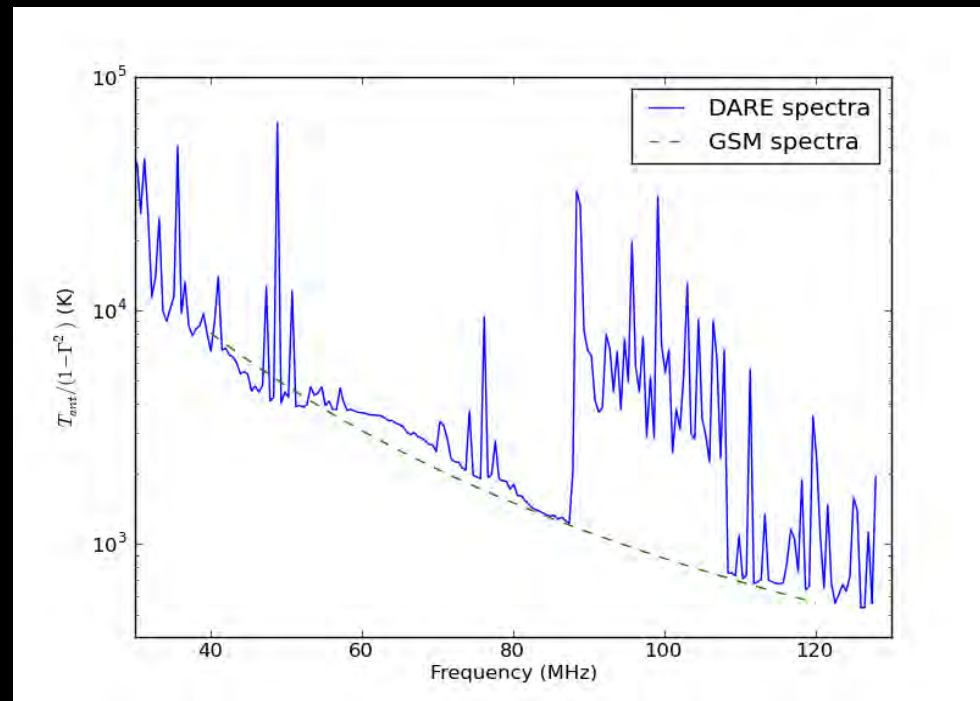
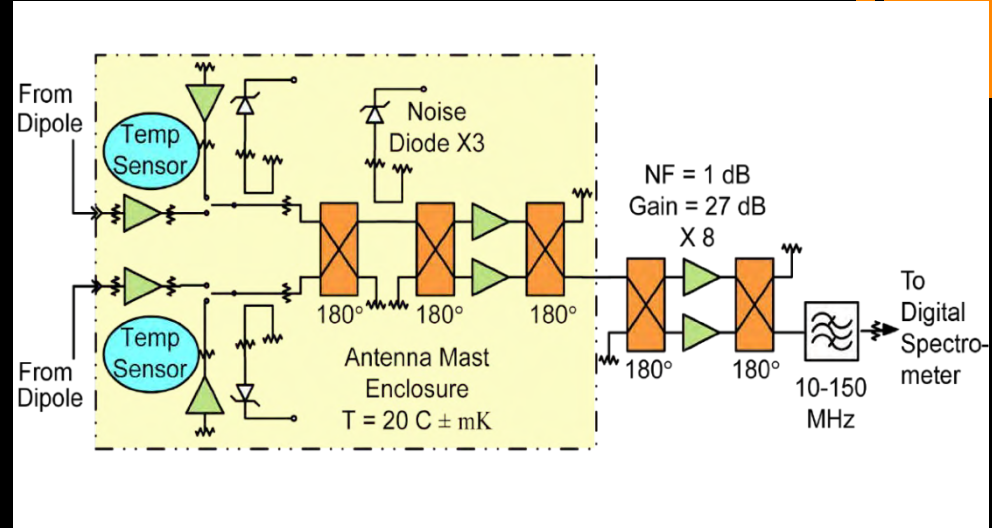
$$T_{sky} \sim T_{Ant} / (1 - \Gamma^2)$$

where Γ is the Reflection coefficient.

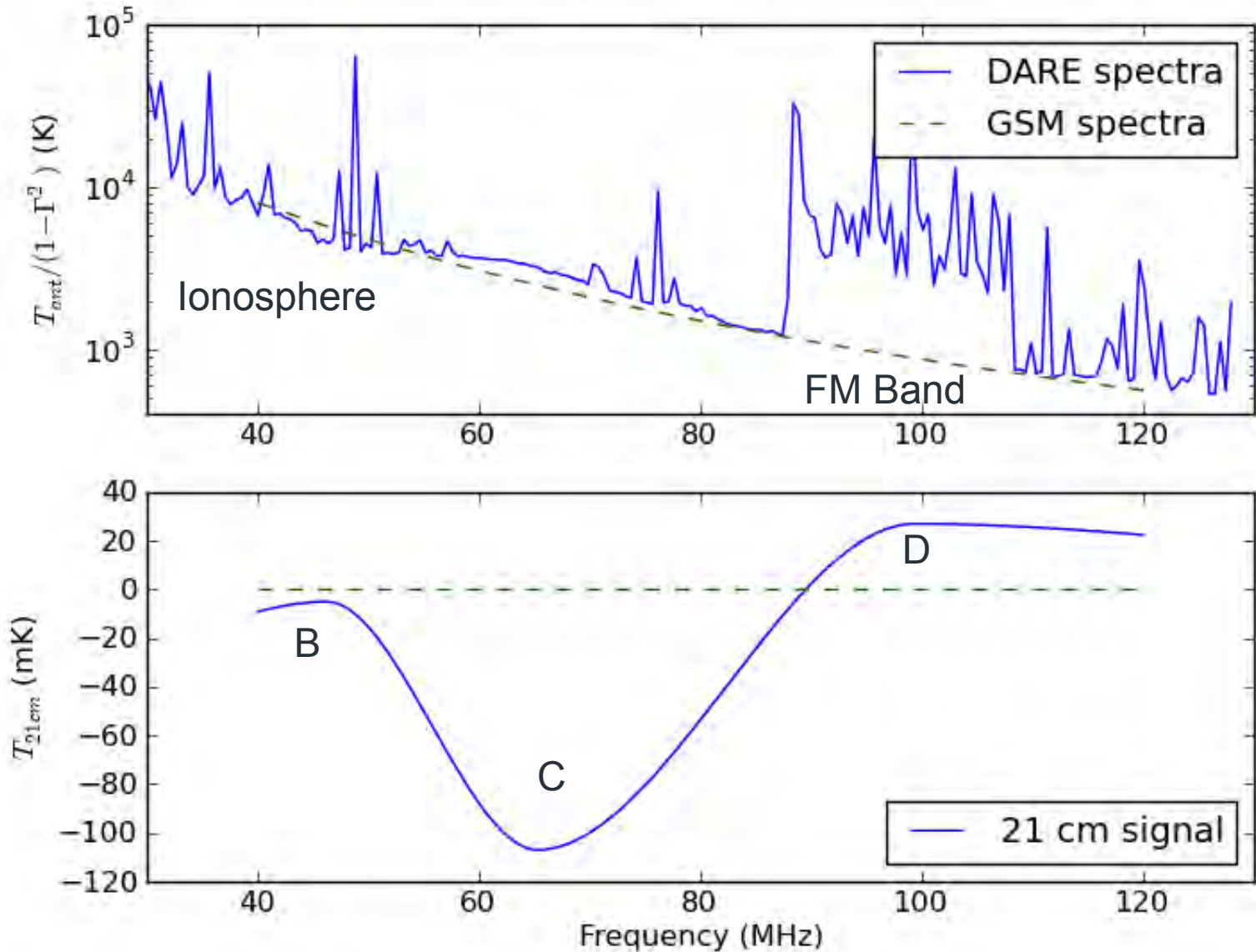
Calibrated spectra shows effects of :

- 1) Radio Frequency Interference (RFI)
- 2) Earth's Ionosphere

These are two major challenges for ground based observations at these frequencies.



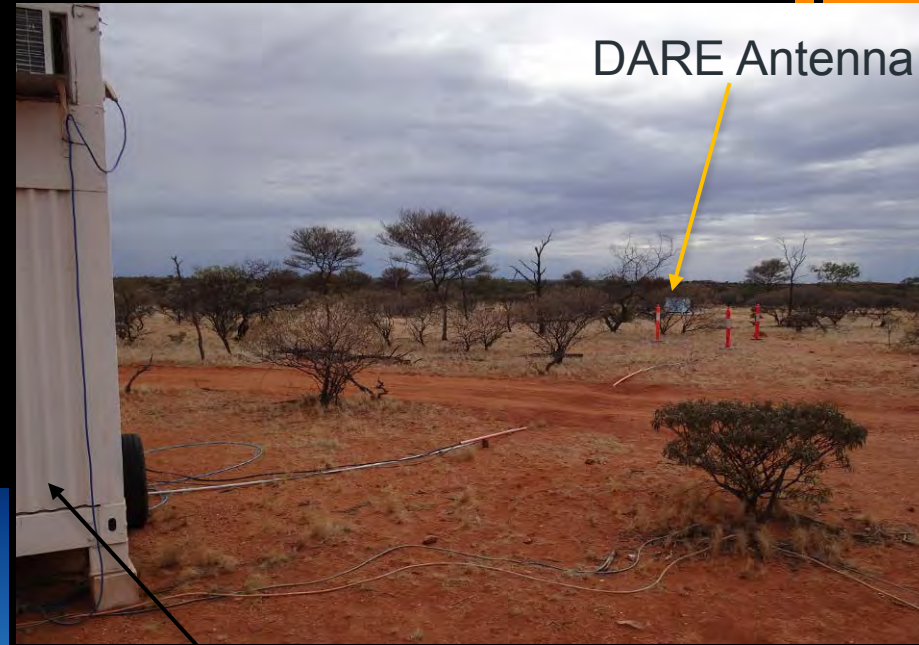
First Results



Galactic synchrotron spectral index ~ 2.7

DARE-ground experiment in Western Australia

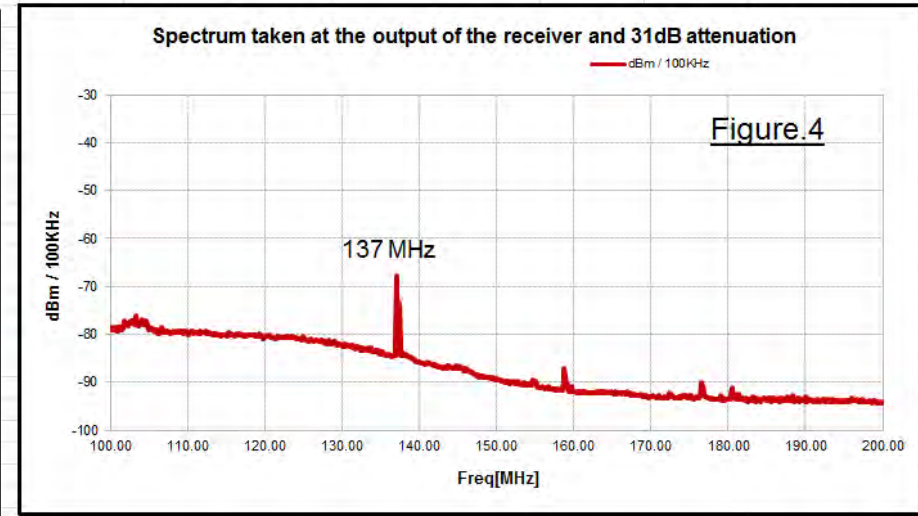
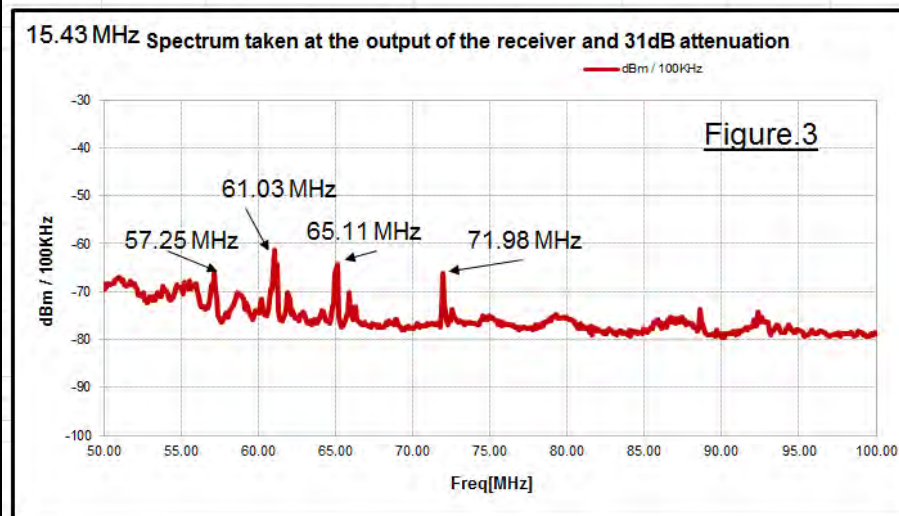
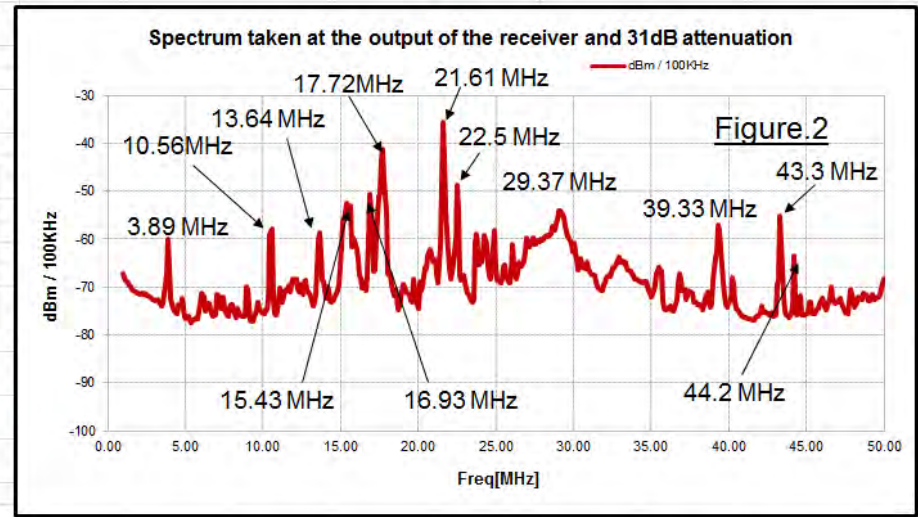
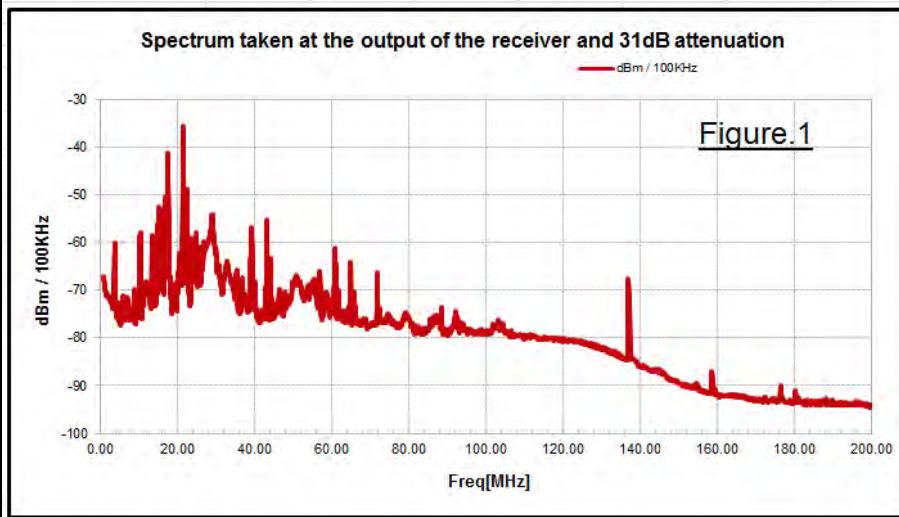
DARE Engineering prototype was deployed on March 21, 2012 at Murchison Radio Observatory (MRO).



EDGS/DARE Instruments HUT



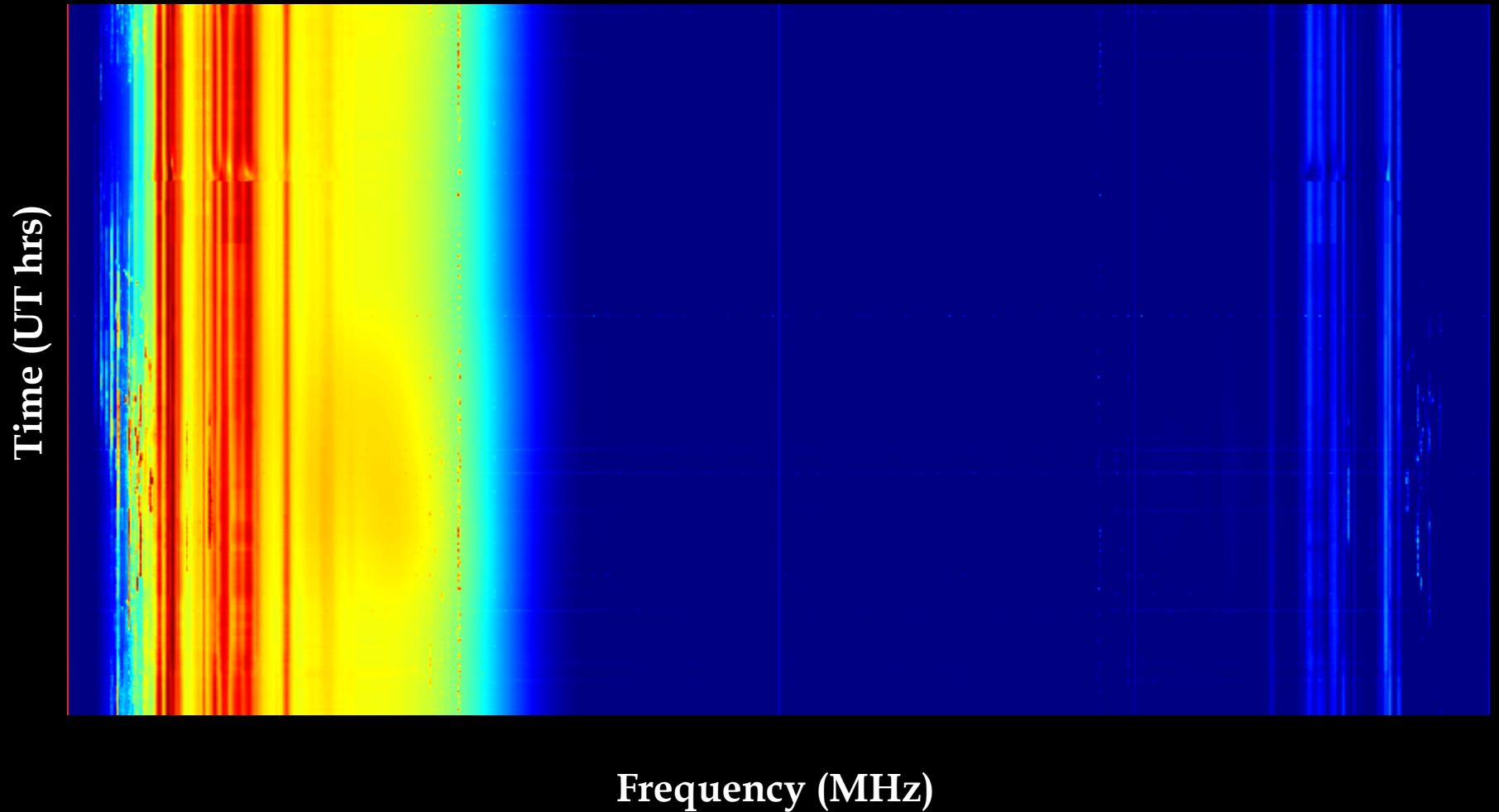
RFI environment at another Radio-Quiet Zone



Strong RFI (@ 20 MHz) caused saturation of the receiver. Modified receiver has been installed.

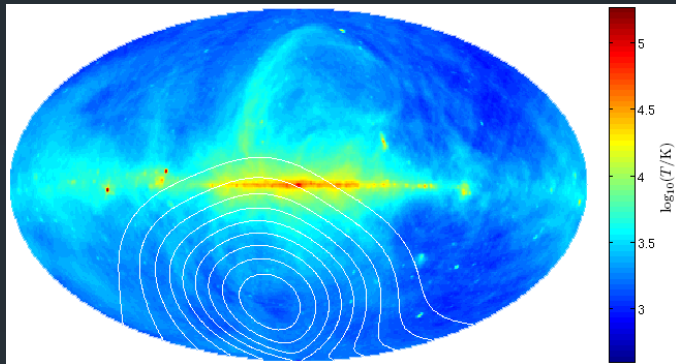
Initial Data at Western Australia

October 30, 2012

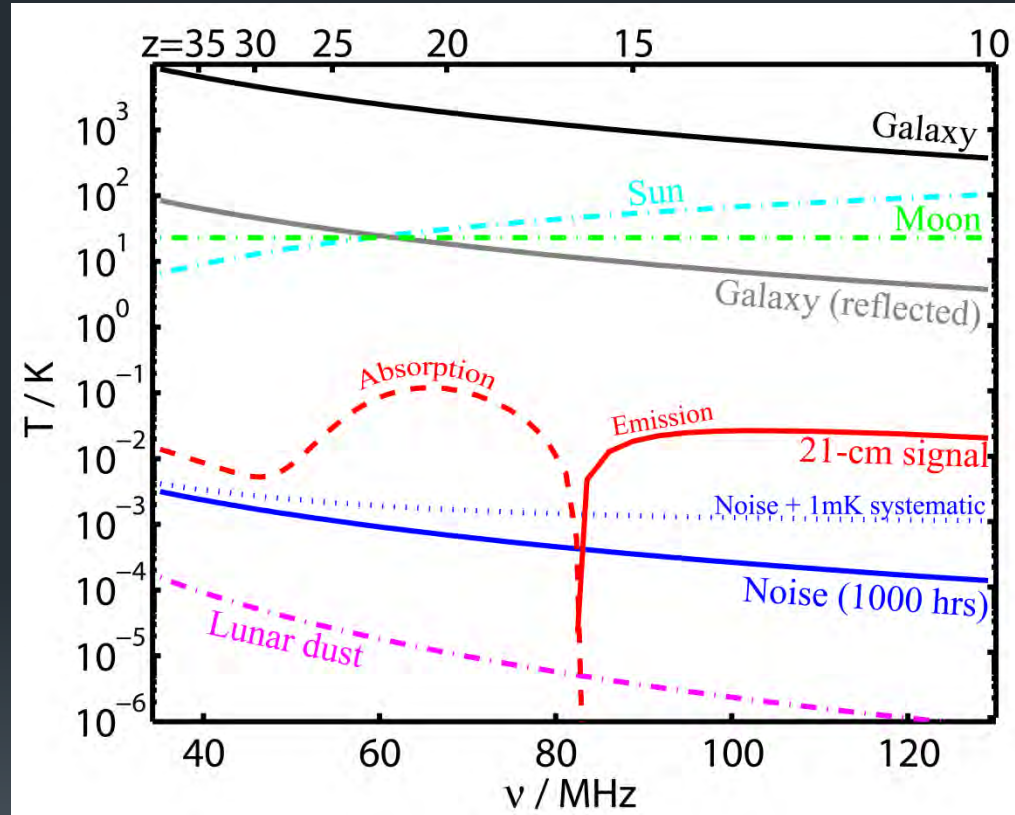


Foregrounds

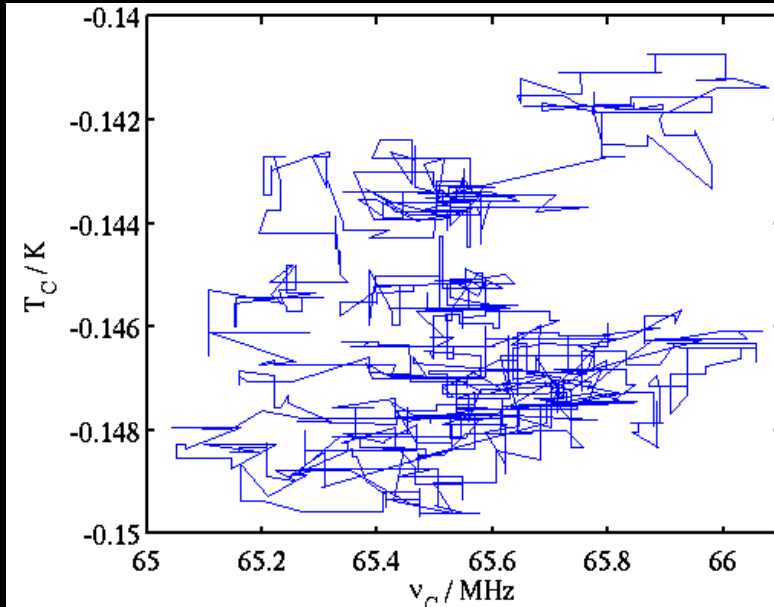
1) Milky Way synchrotron emission + "sea" of extragalactic sources.



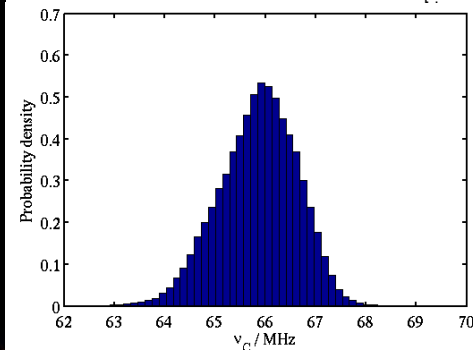
2) Solar system objects: Sun, Jupiter, Moon.



MCMC approach to signal extraction for fiducial DARE mission



- 68% conf.
- 95% conf.



Random walk through parameter space
→ unbiased, random samples of the posterior probability distribution

Positions of turning points B, C and D



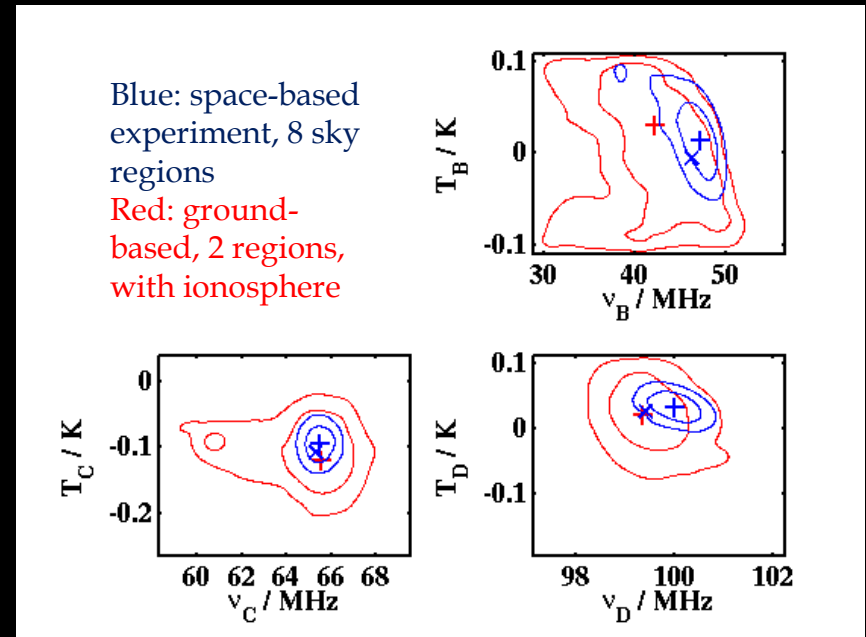
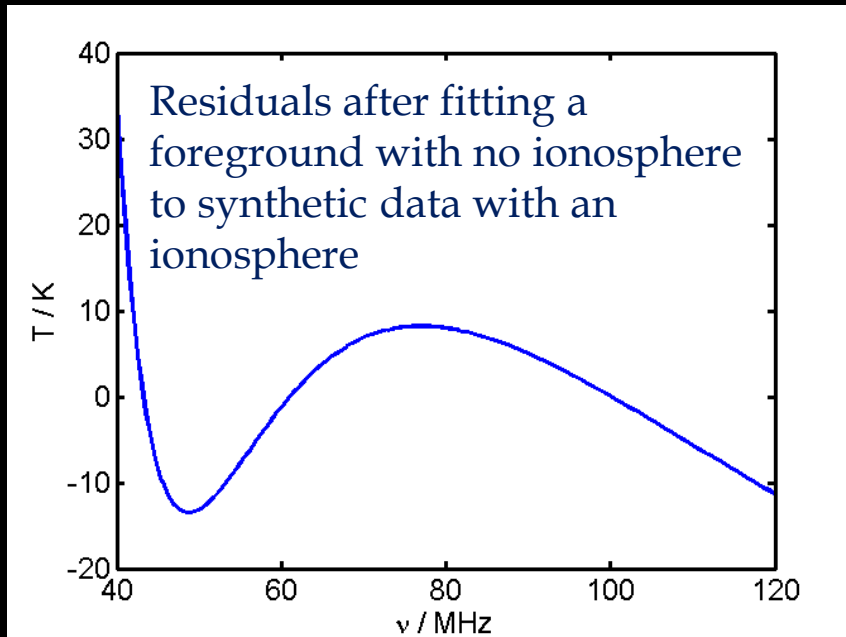
Shape of 21-cm signal

Modeling Earth's Ionosphere

Earth's ionosphere produces a contribution to the spectrum scaling as (frequency)⁻²

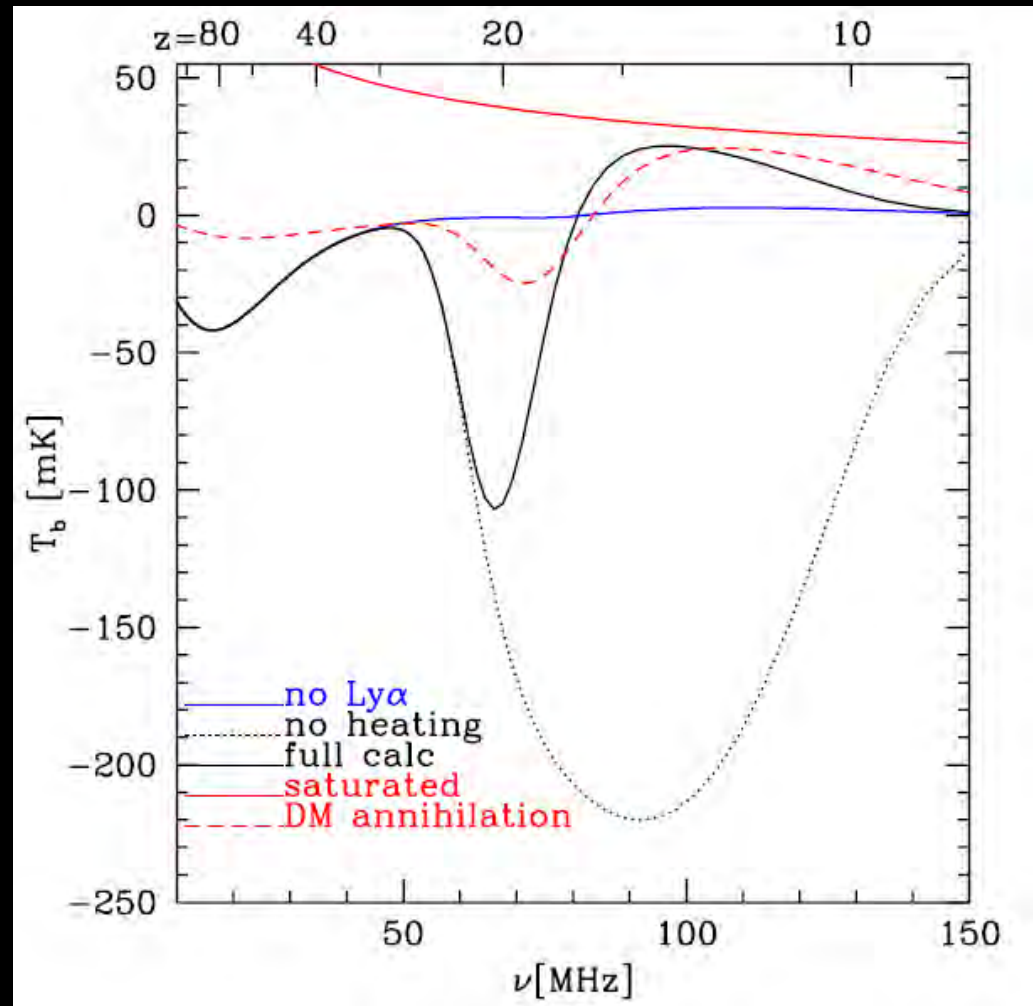
$$T'_{sky} = T_{sky}L_{iono} + (1 - L_{iono})T_e$$

Loss factor $\rightarrow L_{iono} = \frac{1.16 \times 10^{-6}}{\nu_{Hz}^2} \int N(m^{-3}) \nu_{Hz}^c dS (dB)$ Electron temperature



Future work: alternative signal models

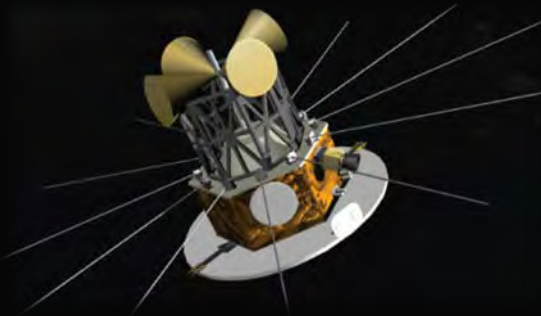
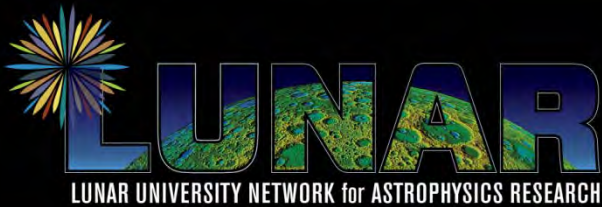
- A wide range of values for the model parameters are allowed by current constraints.
- Alternative models (e.g. including decaying dark matter) may not be described very well by the turning points scheme.
- The nested sampling algorithm will allow us to test how well DARE can select between models with very different shapes.
- We are exploring alternative parametrizations.



Pritchard & Loeb (2010)

Dark Ages Radio Explorer (DARE)

- **DARE is designed to address:**
 - When did the First Stars ignite?
 - When did the first accreting Black Holes turn on?
 - When did Reionization begin?
- **DARE will accomplish this by:**
 - Constructing first sky-averaged spectrum of redshifted 21-cm signal at $11 < z < 35$.
 - Flying spacecraft in lunar orbit & collecting data above lunar farside -- only proven radio-quiet zone in inner solar system.
 - Using biconical dipole antennas with smooth response function & Markov Chain Monte Carlo method to recover spectral *turning points* in the presence of bright foregrounds.
 - Using high heritage spacecraft bus (WISE) & technologies/techniques from EDGES.



See Burns *et al.*, 2012, *Advances in Space Research*, 49, 433.

<http://lunar.colorado.edu/dare/>